TECHNICAL NOTE 2595

DESIGN OF TWO-DIMENSIONAL CHANNELS WITH PRESCRIBED

VELOCITY DISTRIBUTIONS ALONG THE CHANNEL WALLS

II - SOLUTION BY GREEN'S FUNCTION

By John D. Stanitz

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NATIONAL ADVISORY COMMITTEE FOR AERONAUTICS

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SUMMARY

Methods of solution by Green's function are developed for the design of two-dimensional unbranched channels with prescribed velocities as a function of arc length along the channel walls. The methods apply to incompressible and linearized compressible, nonviscous irrotational flow. One numerical example is presented for an accelerating elbow with linearized compressible flow. The elbow shape obtained from the solution by Green's function is the same as that obtained from a solution by relaxation methods for the same prescribed conditions. The time required for the calculations is considerably less for solutions by Green's function.

INTRODUCTION

In this report a general method of design is developed for two-dimensional, compressible or incompressible, nonviscous irrotational flow in unbranched channels with prescribed velocities as a function of arc length along the channel walls. The design of channels with prescribed velocities is important because: (1) boundary-layer separation losses can be avoided by prescribed velocities that do not decelerate rapidly enough to cause separation, (2) shock losses in compressible flow and cavitation in incompressible flow can be avoided by prescribed velocities that do not exceed certain maximum values dictated by these phenomena, and (3) for compressible flow the desired flow rate can be assured by prescribed velocities that do not result in "choke flow" conditions.

In Part I of this report (reference 1) solutions were obtained by relaxation methods. This method of solution results in complete information concerning the distribution of flow conditions throughout the channel and can be used to obtain solutions for incompressible flow and

for two types of compressible flow: the general type with arbitrary value for the ratio of specific heats γ (1.4, for example) and the linearized type with γ equal to -1.0.

In the present report solutions are obtained by Green's function. This method of solution is limited to incompressible and linearized $(\gamma=-1.0)$ compressible flow, but the method is more rapid than relaxation methods, provided information within the channel is not required. The method of solution is developed for the channel walls only although the method can be extended to determine the shape of streamlines within the channel.

The design method reported herein was developed at the NACA Lewis laboratory during 1950 and is part of a doctoral thesis conducted with the advice of Professor Ascher H. Shapiro of the Massachusetts Institute of Technology.

METHOD OF SOLUTION

The design method is developed for two-dimensional channels with prescribed velocities along the channel walls. The prescribed velocity is arbitrary except that stagnation points (zero velocity) cannot be prescribed. This exception limits the design method to unbranched channels. In the present report the method of solution is by Green's function in conjunction with a formula derived (elsewhere) from Green's theorem.

Preliminary Considerations

Assumptions. - The fluid is assumed to be nonviscous and either compressible or incompressible. If the fluid is compressible, the ratio of specific heats γ is assumed to be -1.0, so that the differential equations describing the flow are linear. The flow is assumed to be two-dimensional and irrotational.

Physical plane. - The flow field of the two-dimensional channel is considered to lie in the physical xy-plane where x and y are Cartesian coordinates expressed as ratios of a characteristic length equal to the constant channel width downstream at infinity. (All symbols are defined in appendix A.)

At each point in the channel the velocity vector (fig. 1) has a magnitude Q and a direction θ where Q is the fluid velocity expressed as the ratio of a characteristic velocity equal to the constant channel velocity downstream at infinity. For compressible flow, the velocity Q is related to the velocity q by

$$q = Qq_{d}$$
 (1)

where q is the velocity expressed as a ratio of the stagnation speed of sound and the subscript d refers to conditions downstream at infinity.

Stream function Ψ . - If the condition of continuity is satisfied, a stream function Ψ can be defined such that for incompressible flow

$$d\Psi = \frac{\pi}{2} d\psi \tag{2a}$$

where, from Part I,

$$d\psi = Q dn \tag{2b}$$

where n is distance in the xy-plane measured normal to the streamline and expressed as a ratio of the channel width downstream at infinity. For linearized compressible flow $(\gamma = -1.0)$

$$d\Psi = \frac{\pi}{2} \frac{d\psi^*}{\Delta \psi^*}$$
 (2c)

where, from Part I,

$$d\psi^* = \rho^* q^* dn \tag{2d}$$

and where $\Delta \psi^*$ is the value of ψ^* along the left channel wall when faced in the direction of flow if the value of ψ^* along the right wall is arbitrarily equal to zero. The value of $\Delta \psi^*$ is obtained by integrating equation (2d) across the channel at a position far downstream where flow conditions are uniform

$$\Delta \psi^* = \rho_{\bar{d}}^* q_{\bar{d}}^* \tag{2e}$$

From Part I, $\rho*$ is related to the density ρ , expressed as the ratio of a characteristic density equal to the stagnation density, by

$$\rho^* = k_1 \rho \tag{2f}$$

where

$$k_{1} = \frac{1}{\rho_{a}} \sqrt{\frac{1 - \left(\frac{\rho_{a}q_{a}}{\rho_{b}q_{b}}\right)^{2}}{1 - \left(\frac{q_{a}}{q_{b}}\right)^{2}}}$$
(2g)

in which

$$\rho_{a} = \left(1 - \frac{\gamma - 1}{2} q_{a}^{2}\right)^{\frac{1}{\gamma - 1}} \tag{2h}$$

and

$$\rho_{b} = \left(1 - \frac{\gamma - 1}{2} q_{b}^{2}\right)^{\frac{1}{\gamma - 1}} \tag{2i}$$

where the subscripts a and b refer to quantities related to any two selected values of velocity $(q_a \text{ and } q_b, \text{ respectively})$ for which velocities the densities given by equations (2h) and (2i) are equal to the densities ρ given by equations (2f) and (2g). Also, from Part I, q^* is related to q by

$$q^* = k_2 q \tag{2j}$$

where

$$k_2 = \frac{1}{q_b} \sqrt{\frac{\left(\frac{\rho_a}{\rho_b}\right)^2 - 1}{1 - \left(\frac{\rho_a q_a}{\rho_b q_b}\right)^2}}$$
 (2k)

For each prescribed velocity distribution along the channel walls there are an infinite number of linearized compressible flow solutions, depending on the selected values of \mathbf{q}_a and \mathbf{q}_b in equations (2g) and (2k). However, for values of \mathbf{q}_a and \mathbf{q}_b within the range of \mathbf{q} prescribed along the channel walls (and therefore everywhere in the channel), the solutions, that is, channel shapes, probably differ only in small detail (Part I).

The values of q_a and q_b might, for example, be selected to equal the maximum and minimum values of q (which values of q must occur on the channel walls and are therefore known). Also, the values of q_a and q_b might be selected to equal the upstream and downstream velocities q_u and q_d . In this case the upstream and downstream channel widths would then satisfy continuity for a gas with the correct (arbitrary) value of γ (1.4, for example). If the upstream and downstream velocities are equal, their value and the value of some other velocity (the maximum or minimum velocity, for example) can be selected for q_a and q_b or, if desired, q_a can be equal to q_b so that

$$q_a = q_b = q$$

and

$$\rho_a = \rho_b = \rho$$

and, from Part I,

$$k_{1} = \frac{1}{\rho} \sqrt{\frac{1 - \frac{\gamma + 1}{2} q^{2}}{1 - \frac{\gamma - 1}{2} q^{2}}}$$
 (21)

and

$$k_2 = \sqrt{\frac{1}{1 - \frac{\gamma + 1}{2} q^2}}$$
 (2m)

Equations (2a) and (2c) define the stream function Ψ for incompressible and linearized compressible flow, respectively. For both types of flow Ψ varies from zero along the right side of the channel, when faced in the direction of flow, to $\frac{\pi}{2}$ along the left side of the channel.

Velocity potential Φ . - If the condition of irrotational fluid motion is satisfied, a velocity potential Φ can be defined such that for incompressible flow

$$d\Phi = \frac{\pi}{2} d\Phi \tag{3a}$$

where, from Part I,

$$d\Phi = Q ds$$
 (3b)

where s is distance in the xy-plane measured along the streamlines and expressed as the ratio of channel width downstream at infinity. For linearized compressible flow

$$d\Phi = \frac{\pi}{2} \frac{d\phi^*}{\Delta \psi^*} \tag{3c}$$

where, from Part I,

$$d\phi^* = q^* ds \tag{3d}$$

Equations (3a) and (3c) define the velocity potential Φ for incompressible and linearized compressible flow, respectively.

Outline of design method. - Solutions for two-dimensional flow are boundary-value problems. That is, the solutions depend on known conditions imposed along the boundaries of the problem. In the inverse problem of channel design the geometry of the channel walls in the physical xy-plane is unknown. This unknown geometry apparently precludes the possibility of solving the problem in the physical plane and necessitates the use of some new set of coordinates, that is, a transformed plane, in which to solve the problem. These new coordinates must be such that the geometric boundaries along which the velocities are prescribed are known in the transformed plane. It is also necessary, for the method of solution employed in this report, that the coordinate system of the transformed plane be orthogonal in the physical plane. A set of coordinates that satisfies these requirements is provided by Φ and Ψ , which are orthogonal in the physical xy-plane and for which the geometric boundaries are known constant values of \(\mathbf{Y} \) (equal to $\frac{\pi}{2}$) in the transformed $\Phi\Psi$ -plane. The distribution of velocity as a function of Φ along these boundaries of constant Ψ is known because, if

$$Q = Q(s)$$

or

$$q = q(s)$$

is prescribed, equation (3a) or (3c) integrates to give

$$\Phi = \Phi(s)$$

from which

$$Q = Q(\Phi)$$

or

$$q = q(\Phi)$$

The technique of the proposed channel-design method is therefore to solve for the physical x,y-coordinates of the channel walls in the transformed $\Phi\Psi$ -plane where the prescribed boundary conditions for the two-dimensional flow problem are known.

Channel wall coordinates. - From Part I the distribution of channel wall coordinates x and y along the boundaries of constant Ψ equal to 0 and $\frac{\pi}{2}$ in the transformed $\Phi\Psi$ -plane is given by

$$x = \frac{2}{\pi} \Delta \psi^* \int \frac{\cos \theta}{q^*} d\Phi$$
 (4a)

and

$$y = \frac{2}{\pi} \Delta \psi^* \int \frac{\sin \theta}{q^*} d\Phi \qquad (4b)$$

for linearized compressible flow, and for incompressible flow

$$x = \frac{2}{\pi} \int \frac{\cos \theta}{Q} d\Phi$$
 (5a)

and

$$y = \frac{2}{\pi} \int \frac{\sin \theta}{Q} d\Phi$$
 (5b)

where the constants of integration are selected to give known (specified) values of x or y at one value of Φ along each boundary. Because q^* and Q are known functions of Φ from the prescribed velocity as a function of arc length along the channel walls, the shape of the channel walls in the physical xy-plane is given by equation (4) or (5) if θ is determined as a function of Φ along the channel walls (Ψ equals 0 and $\frac{\pi}{2}$). In this report the solution for θ as a function of Φ along the channel walls in the $\Phi\Psi$ -plane is obtained by Green's function.

Solution by Green's Function

Continuity. - From Part I the continuity equation becomes in the transformed $\Phi\Psi$ -plane

$$\frac{\partial \log_{\rm e} V}{\partial \Phi} + \frac{\partial \Psi}{\partial \Psi} = 0 \tag{6a}$$

where for incompressible flow

$$V = Q (6b)$$

and for linearized compressible flow

$$V = \frac{q^*}{1 + \sqrt{1 + q^{*2}}}$$
 (6c)

Irrotational motion. - From Part I the equation for irrotational motion becomes in the transformed $\Phi\Psi\text{-plane}$

$$\frac{\partial \log_{\mathsf{e}} V}{\partial \Psi} - \frac{\partial \theta}{\partial \Phi} = 0 \tag{7}$$

Integral equation for $\theta(\Phi_O, \Psi_O)$. - From equations (6a) and (7)

$$\frac{\partial \Phi^2}{\partial z^2} + \frac{\partial \Psi^2}{\partial z^2} = 0 \tag{8}$$

so that from appendix B the value of θ at a point (Φ_O,Ψ_O) within, or on, the channel walls in the transformed $\Phi\Psi$ -plane is given by the integral equation

$$\theta(\Phi_{O}, \Psi_{O}) = \frac{-1}{2\pi} \int_{-\infty}^{\infty} \left[\left(G \frac{\partial \log_{e} V}{\partial \Phi} \right)_{\frac{\pi}{2}} - \left(G \frac{\partial \log_{e} V}{\partial \Phi} \right)_{O} \right] d\Phi$$
 (9)

where the subscripts 0 and $\frac{\pi}{2}$ refer to the channel wall boundaries along which Ψ is 0 and $\frac{\pi}{2}$, respectively, and G is the Green's function of the second kind for the channel, which is an infinite strip of width $\frac{\pi}{2}$ extending in the Φ -direction to $\pm \infty$.

Green's function G. - The Green's function of the second kind G for the infinite channel in the $\Phi\Psi$ -plane is given along the channel wall boundaries (Ψ equals 0 and $\frac{\pi}{2}$) by (appendix C)

$$G_{O \text{ or } \frac{\pi}{2}} = -\log_e \left[\cosh^2 (\Phi - \Phi_O) - \cos^2 (\Psi - \Psi_O) \right]$$
 (10)

where (Φ,Ψ) is any point on the channel wall boundary and (Φ_0,Ψ_0) is the point in the channel or on the boundary at which θ is to be determined.

Numerical integration for $\theta(\Phi_0, \Psi_0)$. - From equations (9) and (10)

$$2\pi \ \theta(\Phi_{_{\! O}}, \Psi_{_{\! O}}) \ = \!\! \int_{-\infty}^{\infty} \! \left\{ \frac{\partial \log_{e} \, V}{\partial \Phi} \, \log_{e} \, \left[\cosh^{2} \, (\Phi \! - \! \Phi_{_{\! O}}) \, - \, \sin^{2} \, \Psi_{_{\! O}} \right] \right\}_{\!\! \frac{\pi}{2}} \, \mathrm{d}(\Phi \! - \! \Phi_{_{\! O}}) \ - \, \left[- \, \exp^{2} \, \left(\Phi_{_{\! O}} \right) \, - \, \exp^{2} \, \left(\Phi_{_{\! O}} \right) \, - \, \exp^{2} \, \left(\Phi_{_{\! O}} \right) \, \right] \right\}_{\!\! \frac{\pi}{2}} \, \mathrm{d}(\Phi \! - \! \Phi_{_{\! O}}) \ - \, \left[- \, \exp^{2} \, \left(\Phi_{_{\! O}} \right) \, - \, \exp^{2} \, \left(\Phi_{_{\! O}} \right) \, \right] \right]_{\!\! 0} \, \mathrm{d}(\Phi \! - \! \Phi_{_{\! O}}) \ - \, \left[- \, \exp^{2} \, \left(\Phi_{_{\! O}} \right) \, - \, \exp^{2} \, \left(\Phi_{_{\! O}} \right) \, \right] \right]_{\!\! 0} \, \mathrm{d}(\Phi \! - \! \Phi_{_{\! O}}) \ - \, \left[- \, \exp^{2} \, \left(\Phi_{_{\! O}} \right) \, - \, \exp^{2} \, \left(\Phi_{_{\! O}} \right) \, \right] \right]_{\!\! 0} \, \mathrm{d}(\Phi \! - \! \Phi_{_{\! O}}) \ - \, \left[- \, \exp^{2} \, \left(\Phi_{_{\! O}} \right) \, - \, \exp^{2} \, \left(\Phi_{_{\! O}} \right) \, \right]_{\!\! 0} \, \mathrm{d}(\Phi \! - \! \Phi_{_{\! O}}) \ - \, \left[- \, \Phi_{_{\! O}} \right]_{\!\! 0} \, \mathrm{d}(\Phi \! - \! \Phi_{_{\! O}}) \ - \, \left[- \, \Phi_{_{\! O}} \right]_{\!\! 0} \, \mathrm{d}(\Phi \! - \! \Phi_{_{\! O}}) \ - \, \left[- \, \Phi_{_{\! O}} \right]_{\!\! 0} \, \mathrm{d}(\Phi \! - \! \Phi_{_{\! O}}) \ - \, \left[- \, \Phi_{_{\! O}} \right]_{\!\! 0} \, \mathrm{d}(\Phi \! - \! \Phi_{_{\! O}}) \ - \, \left[- \, \Phi_{_{\! O}} \right]_{\!\! 0} \, \mathrm{d}(\Phi \! - \! \Phi_{_{\! O}}) \ - \, \Phi_{_{\! O}} \ - \, \Phi_{_{\! O}} \ + \, \Phi_{_{$$

$$\int_{-\infty}^{\infty} \left\{ \frac{\partial \log_{e} V}{\partial \Phi} \log_{e} \left[\cosh^{2} (\Phi - \Phi_{o}) - \cos^{2} \Psi_{o} \right] \right\}_{0} d(\Phi - \Phi_{o}) \quad (11)$$

in which the independent variable of integration has been changed from $d\Phi$ to $d(\Phi-\Phi_0)$ so that the origin, for purposes of integration, lies at Φ_0 rather than $\Phi=0$. If for small changes in $(\Phi-\Phi_0)$, that is

for small $\Delta\Phi$, the term $\frac{\partial \log_{\rm e} V}{\partial\Phi}$ may be considered constant and equal to its average value over the interval $\Delta\Phi$, then

$$\frac{\partial \log_{e} V}{\partial \Phi} = \frac{\Delta \log_{e} V}{\Delta \Phi}$$

and equation (11) becomes

$$2\pi\theta(\Phi_{o}, \Psi_{o}) = \sum_{(\Phi - \Phi_{o}) = -\infty}^{\infty} \left\{ \frac{\Delta \log_{e} V}{\Delta \Phi} \int_{(\Phi - \Phi_{o})}^{(\Phi - \Phi_{o}) + \Delta \Phi} \log_{e} \left[\cosh^{2}(\Phi - \Phi_{o}) - \sin^{2} \Psi_{o} \right] d(\Phi - \Phi_{o}) \right\}_{\frac{\pi}{2}} - \frac{\Delta \log_{e} V}{\Delta \Phi} \left\{ \frac{\Delta \log_{e} V}{\Delta \Phi} \int_{(\Phi - \Phi_{o})}^{(\Phi - \Phi_{o}) + \Delta \Phi} \log_{e} \left[\cosh^{2}(\Phi - \Phi_{o}) - \sin^{2} \Psi_{o} \right] d(\Phi - \Phi_{o}) \right\}_{\frac{\pi}{2}} - \frac{\Delta \log_{e} V}{\Delta \Phi} \left\{ \frac{\Delta \log_{e} V}{\Delta \Phi} \int_{(\Phi - \Phi_{o})}^{(\Phi - \Phi_{o}) + \Delta \Phi} \log_{e} \left[\cosh^{2}(\Phi - \Phi_{o}) - \sin^{2} \Psi_{o} \right] d(\Phi - \Phi_{o}) \right\}_{\frac{\pi}{2}} - \frac{\Delta \log_{e} V}{\Delta \Phi} \left\{ \frac{\Delta \log_{e} V}{\Delta \Phi} \int_{(\Phi - \Phi_{o})}^{(\Phi - \Phi_{o}) + \Delta \Phi} \log_{e} \left[\cosh^{2}(\Phi - \Phi_{o}) - \sin^{2} \Psi_{o} \right] d(\Phi - \Phi_{o}) \right\}_{\frac{\pi}{2}} - \frac{\Delta \log_{e} V}{\Delta \Phi} \left\{ \frac{\Delta \log_{e} V}{\Delta \Phi} \right\}_{\frac{\pi}{2}} - \frac{\Delta \log_{e} V}{\Delta \Phi} \left\{ \frac{\Delta \log_{e} V}{\Delta \Phi} \right\}_{\frac{\pi}{2}} - \frac{\Delta \log_{e} V}{\Delta \Phi} \left\{ \frac{\Delta \log_{e} V}{\Delta \Phi} \right\}_{\frac{\pi}{2}} - \frac{\Delta \log_{e} V}{\Delta \Phi} \left\{ \frac{\Delta \log_{e} V}{\Delta \Phi} \right\}_{\frac{\pi}{2}} - \frac{\Delta \log_{e} V}{\Delta \Phi} \left\{ \frac{\Delta \log_{e} V}{\Delta \Phi} \right\}_{\frac{\pi}{2}} - \frac{\Delta \log_{e} V}{\Delta \Phi} \left\{ \frac{\Delta \log_{e} V}{\Delta \Phi} \right\}_{\frac{\pi}{2}} - \frac{\Delta \log_{e} V}{\Delta \Phi} \left\{ \frac{\Delta \log_{e} V}{\Delta \Phi} \right\}_{\frac{\pi}{2}} - \frac{\Delta \log_{e} V}{\Delta \Phi} \left\{ \frac{\Delta \log_{e} V}{\Delta \Phi} \right\}_{\frac{\pi}{2}} - \frac{\Delta \log_{e} V}{\Delta \Phi} \left\{ \frac{\Delta \log_{e} V}{\Delta \Phi} \right\}_{\frac{\pi}{2}} - \frac{\Delta \log_{e} V}{\Delta \Phi} \left\{ \frac{\Delta \log_{e} V}{\Delta \Phi} \right\}_{\frac{\pi}{2}} - \frac{\Delta \log_{e} V}{\Delta \Phi} \left\{ \frac{\Delta \log_{e} V}{\Delta \Phi} \right\}_{\frac{\pi}{2}} - \frac{\Delta \log_{e} V}{\Delta \Phi} \left\{ \frac{\Delta \log_{e} V}{\Delta \Phi} \right\}_{\frac{\pi}{2}} - \frac{\Delta \log_{e} V}{\Delta \Phi} \left\{ \frac{\Delta \log_{e} V}{\Delta \Phi} \right\}_{\frac{\pi}{2}} - \frac{\Delta \log_{e} V}{\Delta \Phi} \left\{ \frac{\Delta \log_{e} V}{\Delta \Phi} \right\}_{\frac{\pi}{2}} - \frac{\Delta \log_{e} V}{\Delta \Phi} \left\{ \frac{\Delta \log_{e} V}{\Delta \Phi} \right\}_{\frac{\pi}{2}} - \frac{\Delta \log_{e} V}{\Delta \Phi} \left\{ \frac{\Delta \log_{e} V}{\Delta \Phi} \right\}_{\frac{\pi}{2}} - \frac{\Delta \log_{e} V}{\Delta \Phi} \left\{ \frac{\Delta \log_{e} V}{\Delta \Phi} \right\}_{\frac{\pi}{2}} - \frac{\Delta \log_{e} V}{\Delta \Phi} \left\{ \frac{\Delta \log_{e} V}{\Delta \Phi} \right\}_{\frac{\pi}{2}} - \frac{\Delta \log_{e} V}{\Delta \Phi} \left\{ \frac{\Delta \log_{e} V}{\Delta \Phi} \right\}_{\frac{\pi}{2}} - \frac{\Delta \log_{e} V}{\Delta \Phi} \left\{ \frac{\Delta \log_{e} V}{\Delta \Phi} \right\}_{\frac{\pi}{2}} - \frac{\Delta \log_{e} V}{\Delta \Phi} \left\{ \frac{\Delta \log_{e} V}{\Delta \Phi} \right\}_{\frac{\pi}{2}} - \frac{\Delta \log_{e} V}{\Delta \Phi} \left\{ \frac{\Delta \log_{e} V}{\Delta \Phi} \right\}_{\frac{\pi}{2}} - \frac{\Delta \log_{e} V}{\Delta \Phi} \left\{ \frac{\Delta \log_{e} V}{\Delta \Phi} \right\}_{\frac{\pi}{2}} - \frac{\Delta \log_{e} V}{\Delta \Phi} \left\{ \frac{\Delta \log_{e} V}{\Delta \Phi} \right\}_{\frac{\pi}{2}} - \frac{\Delta \log_{e} V}{\Delta \Phi} \left\{ \frac{\Delta \log_{e} V}{\Delta \Phi} \right\}_{\frac{\pi}{2}} - \frac{\Delta \log_{e} V}{\Delta \Phi} \left\{ \frac{\Delta \log_{e} V}{\Delta \Phi} \right\}_{\frac{\pi}{2}} - \frac{\Delta \log_{e} V}{\Delta \Phi} \left\{ \frac{\Delta \log_{e$$

$$\sum_{(\Phi-\Phi_{o})=-\infty}^{\infty} \left\{ \frac{\Delta \log_{e} V}{\Delta \Phi} \int_{(\Phi-\Phi_{o})}^{(\Phi-\Phi_{o})+\Delta \Phi} \log_{e} \left[\cosh^{2}(\Phi-\Phi_{o}) - \cos^{2} \Psi_{o} \right] d(\Phi-\Phi_{o}) \right\}_{0}$$
(12)

where the summation sign is understood to mean that the quantity within the braces is summed over the entire range of $(\Phi - \Phi_0)$ between $\pm \infty$.

Equation (12) determines θ at any point in the flow field (channel). For a point (Φ_0,Ψ_0) on the channel walls Ψ_0 is equal to 0 or $\frac{\pi}{2}$ and the integrands in equation (12) become

$$2 \log_{\rm e} \cosh |(\Phi - \Phi_{\rm o})|$$

or

2
$$\log_e \sinh |(\Phi - \Phi_o)|$$

so that equation (12) becomes

$$\pi\theta(\Phi_{O}, \Psi_{O}) = \sum_{(\Phi - \Phi_{O}) = -\infty}^{\infty} \left[\left(\frac{\Delta \log_{e} V}{\Delta \Phi} \Delta I \right)_{\frac{\pi}{2}} - \left(\frac{\Delta \log_{e} V}{\Delta \Phi} \Delta I \right)_{O} \right] \quad (13a)$$

where

$$\Delta I = I_{(\Phi - \Phi_O) + \Delta \Phi} - I_{(\Phi - \Phi_O)}$$
 (13b)

$$I_{\frac{\pi}{2}} = \alpha \quad \text{if} \quad \Psi_{O} = 0$$

$$I_{\frac{\pi}{2}} = \beta \quad \text{if} \quad \Psi_{O} = \frac{\pi}{2}$$

$$I_{O} = \alpha \quad \text{if} \quad \Psi_{O} = \frac{\pi}{2}$$

$$I_{O} = \beta \quad \text{if} \quad \Psi_{O} = 0$$

$$(13c)$$

where

$$\alpha = \pm \int_{0}^{|(\Phi - \Phi_{o})|} \log_{e} \cosh |(\Phi - \Phi_{o})| d|(\Phi - \Phi_{o})|$$
(13d)

$$\beta = \pm \int_{0}^{||(\Phi - \Phi_{o})|} \log_{e} \sinh|(\Phi - \Phi_{o})| d|(\Phi - \Phi_{o})|$$
 (13e)

where the + signs apply for positive values of $(\Phi - \Phi_0)$ and the - signs apply for negative values of $(\Phi - \Phi_0)$. Methods of evaluating α and β are given in appendix D and tabulated values are given for a wide range of $|(\Phi - \Phi_0)|$ in table I. Equation (13a) determines $\theta(\Phi_0, \Psi_0)$ at any point on the channel wall boundaries. Thus from equations (4a) and (4b) or (5a) and (5b) the coordinates for the channel wall shape in the physical xy-plane can be determined.

NUMERICAL PROCEDURE

The numerical procedure for the channel design solution by Green's function is the same, except for minor details, for incompressible and linearized compressible flow. The stepwise procedure is outlined as follows:

(1) For incompressible flow the velocity Q and for linearized compressible flow the velocity q, or which is the same thing the velocity Q and the constant downstream velocity $\mathbf{q_d}$, are specified as functions of arc length along the channel walls

$$Q = Q(s) \tag{14a}$$

$$q = q(s) \tag{14b}$$

where s is arbitrarily equal to 0 at that point along one channel wall where the velocity first begins to vary.

(2) Compute V as a function of s from equations (6b) and (14a) for incompressible flow or from equations (2j), (2k), (6c), and (14b) for linearized compressible flow.

$$V = V(s) \tag{15}$$

(3) Compute Φ as a function of s from equations (3a) and (3b) for incompressible flow or from equations (2e), (3c), (3d), and (14b) for linearized compressible flow. In equation (2e) ρ_d^* is obtained from equations (2f) to (2i). For arbitrary distributions of Q or q equation (3a) or (3c) is integrated numerically using, for example, Simpson's one-third rule. Thus

$$\Phi = \Phi(s) \tag{16}$$

(4) From equations (15) and (16) V and Φ are known functions of s so that

$$V = V(\Phi) \tag{17}$$

Thus V is a known function of Φ along the channel wall boundaries in the transformed $\Phi\Psi\text{-plane}$

- (5) If the prescribed velocity distribution along one wall is different from that along the other, the channel will, in general, turn the flow. This turning angle $\Delta\theta$ is given by equation (E5) in appendix E. If the turning angle is unsatisfactory a new distribution of velocity as a function of s (equations (14a) and (14b)) is prescribed and steps (1) to (5) repeated until the desired value of $\Delta\theta$ is obtained. Equation (E5) is integrated numerically using Simpson's onethird rule, for example, and equation (17).
- (6) The channel wall boundaries are straight parallel lines of constant Ψ equal to 0 and $\frac{\pi}{2}$, and extending to $\pm \infty$ in the Φ -direction. Along these boundaries of constant Ψ a series of equally spaced points are located at each of which the flow direction θ and the physical x,y-coordinates will be determined by numerical integration. In order to use the tables of α and β presented in this report, the point spacing $\Delta\Phi$ must be an even multiple of $\pi/24$. Thus the smallest point spacing $\pi/24$ is equal to 1/12 of the channel width $(\pi/2)$. For

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 $(\Phi - \Phi_0)$ varies with Φ_0 .

a particular prescribed velocity distribution along the channel walls the accuracy of the solution increases, and so does the amount of computing, as the point spacing is reduced. The error for a given point spacing depends on the prescribed velocity distribution and its order of magnitude is given by the leading term of the error series of the formula used for numerical integration (table VIII, reference 2, for example). For the numerical example presented in this report the point spacing $\Delta\Phi$ was $\pi/12$. From equation (17)

$$\frac{\Delta \log_{e} V}{\Delta \Phi} = \frac{(\log_{e} V)_{\Phi + \Delta \Phi} - (\log_{e} V)_{\Phi}}{\Delta \Phi}$$
 (18)

where the subscripts Φ and $\Phi + \Delta \Phi$ refer to adjacent points along the channel boundaries.

- (7) The value of θ at each point (Φ_0,Ψ_0) on the channel wall boundaries is obtained from equation (13a) in which $\frac{\Delta \log_e V}{\Delta \Phi}$ is given by equation (18) and ΔI is given by equations (13b), (13c), and table I. Note that in equation (13a) the origin has been moved to Φ by changing from Φ to $(\Phi \Phi_0)$. Thus the value of $\frac{\Delta \log_e V}{\Delta \Phi}$ for a given value of
- (8) The physical x,y-coordinates at each point on the channel wall boundaries are obtained by the numerical integration of equations (5a) and (5b) for incompressible flow, or equations (4a) and (4b) for linearized compressible flow where $\Delta \psi^*$ is given by equation (2e). The constants of integration in equations (4) and (5) are selected to give known values of x and y at upstream or downstream positions where flow conditions can be considered uniform.

NUMERICAL EXAMPLE

The channel design method of this report has been applied to the design of an elbow for the same conditions as example IV of Part I. The design is for an accelerating elbow with no local decelerations of the prescribed velocities along the channel walls and with linearized compressible flow.

Prescribed velocity distribution. - The prescribed velocity distribution along the channel walls is given by $\mathbf{q}_{\mathbf{d}}$ downstream of the elbow and by \mathbf{Q} as a function of s along the elbow walls. The downstream velocity $\mathbf{q}_{\mathbf{d}}$ is 0.80176. Along the inner wall (with smaller radii) of the elbow the arbitrarily prescribed velocity \mathbf{Q} as a function of arc length s is given by

$$Q = 0.5 (s \le 0)$$

$$Q = \frac{1}{2} + \frac{s^2}{6} - \frac{s^3}{27} (0 \le s \le 3.0)$$

$$Q = 1.0 (s \ge 3.0)$$
(19)

This velocity distribution is plotted in figure 2.

From equations (1), (2j), (3c), and (3d)

$$d\Phi = \frac{\pi}{2} \frac{k_2 q_d}{\Delta \psi *} Qds$$

which together with equation (19) integrates to give

$$\Phi = \frac{\pi}{2} \frac{k_2 q_d}{\Delta \psi^*} (0.5 \text{ s}) \qquad (s \le 0)$$

$$\Phi = \frac{\pi}{2} \frac{k_2 q_d}{\Delta \psi^*} \left(\frac{s}{2} + \frac{s^3}{18} - \frac{s^4}{108} \right) \qquad (0 \le s \le 3.0)$$

$$\Phi = \frac{\pi}{2} \frac{k_2 q_d}{\Delta \psi^*} \quad (-0.75 + s) \qquad (s \ge 3.0)$$

where from equations (2h), (2i), and (2k) the constant k_2 is equal to 1.36332 and from equations (2e) to (2k) the constant $\Delta\psi^*$ is equal to 0.73782. From equations (1), (2j), (6c), (19), and (20) the variation in $\log_e V$ with Φ was obtained and is plotted in figure 3.

The distribution of velocity as a function of arc length is the same for both channel walls, but, as indicated in figure 3, the distribution on the outer wall (larger radii in xy-plane) is shifted in the positive Φ -direction an amount equal to $\frac{5}{3}\pi$ relative to the distribution on the inner wall. Thus, a velocity difference exists on the two walls at equal values of Φ in the interval $0 \le \Phi \le 3.333\pi$, as shown in figure 3. The greater this difference in velocity or the greater the range in Φ over which velocity differences exist, the greater is the elbow turning angle. For the prescribed velocity distribution given in figures 2 and 3 the elbow turning angle given by equation (E5) is -104.08°.

Results. - The elbow design resulting from the prescribed velocities given in figures 2 and 3 is plotted in figure 4. As indicated in table II the contour of this elbow is very nearly the same as that obtained by relaxation methods for linearized compressible flow with the same prescribed conditions (example IV, Part I).

The solution obtained by Green's function (Part II) required one experienced computer 3 days whereas the solution by relaxation methods (Part I) required about 10 days. The relaxation solutions provide additional information, such as the distribution of velocity across the channel, but for the most part this additional information is of secondary importance and the design of channels by Green's function is more rapid and therefore to be preferred over the design by relaxation methods.

SUMMARY OF RESULTS

Methods of solution by Green's function are developed for the design of two-dimensional unbranched channels with prescribed velocities as a function of arc length along the channel walls. The methods apply to incompressible and linearized-compressible, nonviscous irrotational flow. One numerical example is presented for an accelerating elbow with linearized compressible flow. The elbow shape obtained from the solution by Green's function is the same as that obtained from a solution by relaxation methods for the same prescribed conditions. The time required for the calculations was considerably less for the solution by Green's function.

Lewis Flight Propulsion Laboratory
National Advisory Committee for Aeronautics
Cleveland, Ohio, September 6, 1951

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APPENDIX A

SYMBOLS

The follo	wing symbols are used in this report:
B ₁ ,B ₃ ,	Bernoulli's numbers
c	constant, equation (B3)
G	Green's function of the second kind, equations (B2) and (10)
I ·	integral (α or β)
k ₁	coefficient, equation (2g)
k ₂	coefficient, equation (2k)
1	length of closed boundary
n	distance in xy-plane measured normal to direction of flow (expressed as ratio of characteristic length equal to channel width downstream at infinity)
Q	velocity (expressed as ratio of characteristic velocity equal to constant channel velocity downstream at infinity)
Q	velocity (expressed as ratio of stagnation speed of sound)
q*	velocity used in linearized compressible flow and related to q by equation (2j)
r	distance from any point in $\Phi\Psi$ -plane to point (Φ_0,Ψ_0) at which logarithmic singularity exists
s	distance in xy-plane measured along direction of flow (expressed as ratio of characteristic length equal to channel width downstream at infinity)
v	velocity parameter defined by equations (6b) and (6c) for incompressible and linearized compressible flow, respectively

w,w ₁ ,w ₂	complex functions defined by equations (C3), (Cla), and (C2a), respectively
х,у	Cartesian coordinates in physical plane (expressed as ratios of characteristic length equal to channel width downstream at infinity)
Z .	complex coordinate, equation (Clb)
z	conjugate of z
α	integral, equation (13d)
β	integral, equation (13e)
Υ	ratio of specific heats
Δ	finite increment
θ	flow direction in physical xy-plane (measured in counter-clockwise direction from positive x-axis)
Δθ	channel turning angle, equation (El)
ρ	density (expressed as ratio of stagnation density)
ρ*	density used in linearized compressible flow and related to ρ by equation (2f)
Φ .	velocity potential used as Cartesian coordinate in transformed $\Phi\Psi\text{-plane}$ and related to ϕ or ϕ^{\bigstar} by equation (3a) or (3c), respectively
φ and φ*	velocity potential for incompressible and linearized compressible flow, respectively, equations (3b) and (3d)
Ψ	stream function used as Cartesian coordinate in transformed $\Phi\Psi\text{-plane}$ and related to ψ or ψ^{\bigstar} by equation (2a) or (2c), respectively
ψ and ψ*	stream function for incompressible and linearized compressible flow, respectively, equations (2b) and (2d)
△ψ*	boundary value of ψ^* , for linearized compressible flow, along left channel wall when faced in the direction of flow, equation (2e)

```
any harmonic function in $\P$-plane
ധ
Subscripts:
                   quantities related to any two selected values of velocity
a,b
                      (\mathbf{q}_{\mathrm{a}} and \mathbf{q}_{\mathrm{b}}, respectively) for which densities given
                     by equations (2h) and (2i) are equal to densities given by equations (2f) and (2g)
                   conditions downstream at infinity
                   point in \Phi\Psi-plane at which \theta is determined
                   conditions upstream at infinity
                   point at (\Phi - \Phi_0) on either channel wall boundary
(\Phi - \Phi_0)
                   point at \left[ (\Phi - \Phi_0) + \Delta \Phi \right] on either channel wall boundary
(\Phi - \Phi_0) + \Delta \Phi
                   boundary along which \ \Psi equals 0
0
                   boundary along which \ \Psi$ equals
```

APPENDIX B

INTEGRAL EQUATION FOR $heta(\Phi_{_{\! O}},\Psi_{_{\! O}})$

If the distribution of the angle $\theta(\Phi, \Psi)$ in the transformed $\Phi\Psi$ -plane is harmonic, that is, satisfies equation (8) within and on the channel walls (Ψ equals 0 and $\frac{\pi}{2}$), then from Green's theorem and the theorem of mean value it can be shown that the value of θ at a point (Φ_0, Ψ_0) within (or on) the channel walls is given by (reference 3, p. 204, for example)

$$\theta(\Phi_{O}, \Psi_{O}) = \frac{1}{2\pi} \left[\int_{-\infty}^{\infty} \left(\theta \frac{\partial G}{\partial \Psi} - G \frac{\partial \theta}{\partial \Psi} \right)_{O} d\Phi - \int_{\infty}^{-\infty} \left(-\theta \frac{\partial G}{\partial \Psi} + G \frac{\partial \theta}{\partial \Psi} \right)_{\frac{\pi}{2}} d\Phi \right]$$
(B1)

where the two integrals on the right side of equation (Bl) represent the line integral around the channel walls in the counterclockwise direction with the signs adjusted so that $\frac{\partial}{\partial \Psi}$ represents the inner normal to the path of integration.

The function $G(\Phi,\Psi)$ in equation (B1) is of the form (reference 3, p. 204).

$$G(\Phi, \Psi) = \log_{e} \frac{1}{r} + \omega(\Phi, \Psi)$$
 (B2)

where r is the distance from any point (Φ,Ψ) to the point (Φ_O,Ψ_O) and where $\omega(\Phi,\Psi)$ is an arbitrary function that is harmonic within and on the channel walls. (Thus from equation (B2), $G(\Phi,\Psi)$ is harmonic within and on the channel walls except at the point (Φ_O,Ψ_O) where a logarithmic singularity exists.) Because the harmonic function $\omega(\Phi,\Psi)$ is arbitrary, the function $G(\Phi,\Psi)$ can be selected so that along the channel wall boundaries $(\Psi$ equals 0 and $\frac{\pi}{2})$ $\frac{\partial G}{\partial \Psi}$ is a constant c given by the following equation (obtained from notes presented by Tamarkin and Feller in the 1941 Summer Session for Advanced Instruction and Research in Mechanics at Brown Univ.):

$$c = \frac{2\pi}{l}$$
 (B3)

where l is the length of the path along which the line integral is taken. For the path under consideration l is infinite and therefore

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 $G(\Phi,\Psi)$ can be selected so that $\frac{\partial G}{\partial \Psi}$ is zero along the channel walls. A function with this property is called a Green's function of the second kind. Equation (B1) becomes

$$\theta(\Phi_{O}, \Psi_{O}) = \frac{1}{2\pi} \int_{-\infty}^{\infty} \left[\left(G \frac{\partial \theta}{\partial \Psi} \right)_{\pi} - \left(G \frac{\partial \theta}{\partial \Psi} \right)_{O} \right] d\Phi$$

or, combined with equation (6a)

$$\theta(\Phi_{O}, \Psi_{O}) = \frac{-1}{2\pi} \int_{-\infty}^{\infty} \left[\left(g \frac{\partial \log_{e} V}{\partial \Phi} \right)_{\frac{\pi}{2}} - \left(g \frac{\partial \log_{e} V}{\partial \Phi} \right)_{O} \right] d\Phi$$
 (9)

Along the channel walls $\frac{\partial \log_e V}{\partial \Phi}$ is known from the prescribed velocity distribution so that, after the proper Green's function G has been determined (appendix C), equation (9) determines the value of θ at any point (Φ_O, Ψ_O) . The value of $\theta(\Phi_O, \Psi_O)$ given by equation (9) can be adjusted by an arbitrary constant of integration to give a specified value of θ at one point in the flow field.

APPENDIX C

GREEN'S FUNCTION OF SECOND KIND

From appendix B Green's function of the second kind G satisfies the condition

$$\frac{9\pi}{9G} = 0$$

along the channel walls, which are straight and parallel boundaries (Ψ equals 0 and $\frac{\pi}{2}$) extending to $\pm\infty$ in the Φ -direction, and satisfies the equation

$$\frac{\partial \Phi_{S}}{\partial g^{2}} + \frac{\partial \Psi_{S}}{\partial g^{2}} = 0$$

everywhere in the channel except at the point (Φ_O,Ψ_O) where G has a logarithmic pole. For these conditions the Green's function G can be obtained by analogy from the velocity potential for incompressible flow into a point sink at (Φ_O,Ψ_O) between straight parallel bound-

aries at Ψ equal to 0 and $\frac{\pi}{2}$. The logarithmic pole for G at (Φ_O,Ψ_O) corresponds to the point sink and the condition $\frac{\partial G}{\partial \Psi}=0$ at the boundaries corresponds to zero velocity, that is, no flow normal to the boundaries.

The velocity potential for fluid flow with the boundary conditions just described is obtained from two infinite series of point sinks with the sinks of each series spaced π distance apart in the Ψ -direction and the two series arranged by the method of images in such a manner that no flow crosses the boundaries, that is $\frac{\partial G}{\partial \Psi} = 0.$ This arrangement of point sinks is shown in figure 5.

The complex function w_1 for the first infinite series of point sinks is given by (reference 4, p. 112, for example)

$$w_1 = -\log_e \sinh(z-z_0)$$
 (Cla)

where

$$z = \Phi + i\Psi$$
 (Clb)

The complex function w_2 for the second infinite series of point sinks (mirror image of the first series in order to prevent flow across the boundaries Ψ equals 0 and $\frac{\pi}{2}$) is given by

$$w_2 = -\log_e \sinh (z - \overline{z}_0)$$
 (C2a)

where

$$\overline{z} = \Phi - i\Psi \tag{C2b}$$

The complex function w for the combined flow becomes from equations (Cla) to (C2b)

$$w = w_1 + w_2 = -\log_e \sinh \left[(\Phi - \Phi_o) + i (\Psi - \Psi_o) \right] -$$

$$\log_{e} \sinh \left[\left(\Phi - \Phi_{O} \right) + i \left(\Psi + \Psi_{O} \right) \right] \tag{C3}$$

The Green's function of the second kind G corresponds to the velocity potential for the incompressible flow and is therefore given by the real part of equation (C3)

$$G = -\frac{1}{2} \log_e \left[\cosh^2 (\Phi - \Phi_0) - \cos^2 (\Psi - \Psi_0) \right] \left[\cosh^2 (\Phi - \Phi_0) - \cos^2 (\Psi + \Psi_0) \right]$$
(C4)

But along the channel walls Ψ is equal to 0 or $\frac{\pi}{2}$ so that

$$\cos^2 (\Psi + \Psi_0) = \cos^2 (\Psi - \Psi_0)$$

and equation (C4) becomes

$$G_{O \text{ or }} \frac{\pi}{2} = -\log_e \left[\cosh^2 \left(\Phi - \Phi_O \right) - \cos^2 \left(\Psi - \Psi_O \right) \right]$$
 (10)

Equation (10) gives the Green's function of the second kind along the channel walls (straight parallel lines of constant Ψ equal to 0 and $\frac{\pi}{2}$ and extending to $\pm\infty$ in the Φ -direction).

APPENDIX D

EVALUATION OF α AND β

Several techniques, depending on the magnitude of the upper limit $|\langle\Phi-\Phi_O\rangle|$, were used to evaluate the integrals α and β given by equations (13d) and (13e). Each integral is treated separately in this appendix and the values of $(\Phi-\Phi_O)$ for the upper limit $|\langle\Phi-\Phi_O\rangle|$ are considered positive. For negative values of $(\Phi-\Phi_O)$ the magnitudes of I (that is, of α or β) are equal for corresponding values of $|\langle\Phi-\Phi_O\rangle|$ but opposite in sign. As a result the values of ΔI have the same sign.

Integral a

Small and medium values of $(\Phi - \Phi_0)$. - For small and medium values of the upper limit of integration $(\Phi - \Phi_0)$ in equation (13d), that is, for $0 \le (\Phi - \Phi_0) \le 60 \pi/24$, the integral α is evaluated by Simpson's one-third rule using increments of $(\Phi - \Phi_0)$ equal to $\pi/48$.

Large values of $(\Phi - \Phi_0)$. - For large values of $(\Phi - \Phi_0)$ that is for $(\Phi - \Phi_0) > 60 \pi/24$, the integrand in equation (13d) becomes

$$\log_e \cosh (\Phi - \Phi_0) \approx (\Phi - \Phi_0) - \log_e 2$$
 (D1)

so that equation (13d) becomes

$$\alpha \approx \int_{0}^{60\pi/24} \log_{e} \cosh (\Phi - \Phi_{o}) d(\Phi - \Phi_{o}) + \int_{60\pi/24}^{(\Phi - \Phi_{o})} [(\Phi - \Phi_{o}) - \log_{e} 2] d(\Phi - \Phi_{o})$$

$$\approx 25.809782 + \left[\frac{(\Phi - \Phi_{o})^{2}}{2} - 0.693147 (\Phi - \Phi_{o}) - 25.398552 \right]$$

$$\approx 0.411230 - 0.693147 (\Phi - \Phi_{o}) + \frac{1}{2} (\Phi - \Phi_{o})^{2}$$
(D2)

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Equation (D2) gives values of α for values of $(\Phi-\Phi_0)$ equal to or greater than $60\pi/24$. Values of the integral α are tabulated in table I for a range of $|(\Phi-\Phi_0)|$ between 0 and $100\pi/24$ in increments of $\pi/24$. For negative values of $(\Phi-\Phi_0)$ the sign of α is negative.

Integral B

Small values of $(\Phi - \Phi_O)$. - For $(\Phi - \Phi_O)$ equal to zero the integrand of equation (13e) becomes minus infinity so that Simpson's one-third rule cannot be used to evaluate β in this region of $(\Phi - \Phi_O)$, as was done for α . However, equation (13e) integrates by parts to give

$$\int_{O}^{\left(\Phi-\Phi_{O}\right)}\log_{e}\,\sinh\,\left(\Phi-\Phi_{O}\right)\,\mathrm{d}(\Phi-\Phi_{O})$$

=
$$(\Phi - \Phi_O)$$
 log_e sinh $(\Phi - \Phi_O)$ -

$$\int_{0}^{(\Phi-\Phi_{o})} (\Phi-\Phi_{o}) \operatorname{ctnh} (\Phi-\Phi_{o}) \operatorname{d}(\Phi-\Phi_{o})$$
(D3)

where the integrand $(\Phi - \Phi_0)$ ctnh $(\Phi - \Phi_0)$ on the right side of equation (D3) can be expanded in the following series form:

$$(\Phi - \Phi_{O}) \text{ ctnh } (\Phi - \Phi_{O}) = 1 + \frac{2^{2}B_{1} (\Phi - \Phi_{O})^{2}}{2!} - \frac{2^{4}B_{3} (\Phi - \Phi_{O})^{4}}{4!} + \frac{2^{6}B_{5} (\Phi - \Phi_{O})^{6}}{6!} - \frac{2^{4}B_{5} (\Phi - \Phi_{O})^{4}}{6!} + \frac{2^{6}B_{5} (\Phi - \Phi_{O})^{6}}{6!} - \frac{2^{4}B_{5} (\Phi - \Phi_{O})^{4}}{6!} + \frac{2^{6}B_{5} (\Phi - \Phi_{O})^{6}}{6!} - \frac{2^{4}B_{5} (\Phi - \Phi_{O})$$

$$\frac{2^{8}B_{7} (\Phi - \Phi_{0})^{8}}{8!} + \frac{2^{10}B_{9} (\Phi - \Phi_{0})^{10}}{10!} - \frac{2^{12}B_{11} (\Phi - \Phi_{0})^{12}}{12!} + \dots$$
 (D4)

where B_1 , B_3 , and so forth, are called Bernoulli's numbers (reference 5, p. 90, for example). From equations (D3) and (D4)

$$\beta = (\Phi - \Phi_{o}) \log_{e} \sinh (\Phi - \Phi_{o}) - (\Phi - \Phi_{o}) - \frac{(\Phi - \Phi_{o})^{3}}{9} + \frac{(\Phi - \Phi_{o})^{5}}{225} - \frac{2(\Phi - \Phi_{o})^{7}}{6615} + \frac{(\Phi - \Phi_{o})^{7}}{6615}$$

$$\frac{(\Phi - \Phi_0)^9}{42,525} - \frac{2(\Phi - \Phi_0)^{11}}{1,029,105} + \frac{1382}{8,300,667,375} - \dots$$
 (D5)

Equation (D5) was used to obtain $\,\beta\,$ as a function of $\,(\Phi\!-\!\Phi_{\!_{\rm O}})\,$ for $0\leq (\Phi\!-\!\Phi_{\!_{\rm O}})\leq 8\pi/24$.

Medium values of $(\Phi - \Phi_0)$. - For medium values of the upper limit of integration $(\Phi - \Phi_0)$ in equation (13e), that is, for $8\pi/24 < (\Phi - \Phi_0) \le 60\pi/24$, the integral β is evaluated by Simpson's one-third rule as was done for α .

Large values of $(\Phi - \Phi_0)$. - For large values of $(\Phi - \Phi_0)$, that is, for $(\Phi - \Phi_0) > 60\pi/24$, the integrand in equation (13e) becomes

$$\log_e \sinh (\Phi - \Phi_O) \approx (\Phi - \Phi_O) - \log_e 2$$
 (D6)

so that equation (13e) becomes

$$\beta \approx \int_{0}^{60\pi/24} \log_{e} \sinh (\Phi - \Phi_{o}) d(\Phi - \Phi_{o}) +$$

$$\int_{60\pi/24}^{(\Phi-\Phi_{o})} \left[(\Phi-\Phi_{o}) - \log_{e} 2 \right] d(\Phi-\Phi_{o})$$

≈ 24.576082 +
$$\left[\frac{(\Phi - \Phi_0)^2}{2} - 0.693147 (\Phi - \Phi_0) - 25.398552\right]$$

$$=$$
 0.822470 - 0.693147 ($Φ$ - $Φ$ _o) + $\frac{1}{2}$ ($Φ$ - $Φ$ _o)² (D7)

Equation (D7) gives values of β for values of $(\Phi - \Phi_0)$ equal to or greater than $60\pi/24$. Values of the integral β are tabulated in table I for a range of $|(\Phi - \Phi_0)|$ between 0 and $100\pi/24$ in increments of $\pi/24$. For negative values of $(\Phi - \Phi_0)$ the sign of β changes.

APPENDIX E

CHANNEL TURNING ANGLE

If the prescribed velocity distribution along one channel wall differs from the distribution along the other wall, then in general the channel deflects an amount $\Delta\theta$, which is the difference in flow direction far downstream and far upstream of the region in which the prescribed velocity distribution varies. Thus,

$$\Delta\theta = \theta_{\rm d} - \theta_{\rm n} \tag{E1}$$

For large values of $|(\Phi-\Phi_0)|$ such as occur far upstream and far downstream of the region in which the prescribed velocity varies along the channel walls

$$\cosh^2 (\Phi - \Phi_0) >> \cos^2 (\Psi - \Psi_0)$$

so that from equation (10)

$$G_{O} = G_{\frac{\pi}{2}} = 2 \left[\left| \left(\Phi - \Phi_{O} \right) \right| - \log_{e} 2 \right]$$
 (E2)

Far upstream $\Phi_{\bigcirc} < \Phi$ so that

$$|(\Phi - \Phi_{O})| = (\Phi - \Phi_{O})$$

and because V is harmonic

$$\int_{-\infty}^{\infty} \left[\left(\frac{\partial \log_{e} V}{\partial \Phi} \right)_{\underline{\pi}} - \left(\frac{\partial \log_{e} V}{\partial \Phi} \right)_{0} \right] d\Phi = 0$$

so that equation (E2) substituted in equation (9) gives

$$\theta_{\rm u} = \frac{1}{\pi} \int_{-\infty}^{\infty} \Phi \left[\left(\frac{\partial \log_{\rm e} V}{\partial \Phi} \right)_{\frac{\pi}{2}} - \left(\frac{\partial \log_{\rm e} V}{\partial \Phi} \right)_{0} \right] d\Phi$$
 (E3)

Likewise, far downstream $\Phi_{O} > \Phi$ so that

$$|(\Phi - \Phi_{O})| = - (\Phi - \Phi_{O})$$

and equation (E2) substituted in equation (9) gives

$$\theta_{d} = \frac{-1}{\pi} \int_{-\infty}^{\infty} \Phi \left[\left(\frac{\partial \log_{e} V}{\partial \Phi} \right)_{\underline{\pi}} - \left(\frac{\partial \log_{e} V}{\partial \Phi} \right)_{0} \right] d\Phi$$
 (E4)

From equations (E1), (E3), and (E4)

$$\Delta\theta = \frac{-2}{\pi} \int_{-\infty}^{\infty} \Phi \left[\left(\frac{\partial \log_{e} V}{\partial \Phi} \right)_{\frac{\pi}{2}} - \left(\frac{\partial \log_{e} V}{\partial \Phi} \right)_{0} \right] d\Phi$$
 (E5)

Equation (E5) determines the channel turning angle $\Delta\theta$.

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TABLE I - TABULATED VALUES OF THE INTEGRALS α AND β FOR A RANGE OF $|(\Phi-\Phi_0)|$ [Computational methods given in appendix D.] NACA

F		-	. methods 8			C.NACA	
14 11	a ⁽¹⁾	ΔΙ =	= Δα ·	_B (1)	ΔΙ =	Дβ	
(Φ-Φ _O)	α, ,	Δηα	Δρα	β\-′	Δηβ	Δ2β	
		_	(ΔΦ=2π/24)		(ΔΦ=π/24)	(ΔΦ=2π/24)	
		(22-11) 2-1	(22 23,,				
0	0	0.000373		0	-0.396937	0.022000	
1(π/24)	0.000373	.002597	0.002970	-0.396937	214723	-0.611660	
2(π/24)	.002970	.006972		611660	144746		
3(π/24)	.009942	.013359	.020331	756406	- 098045	242791	
4(π/24)	.023301			854451	062035		
5(π/2 4)	.044875	.021574	.052976	916486	032033	094079	
$6(\pi/24)$.076277	.031402		948530	005816		
7(π/24)	.118897	.042620	.097632	954346		.012092	
8(π/24)	.173909	.055012		936438	.017908		
$9(\pi/24)$.242283	.068374	.150903	896542	.039896	.100542	
10(π/24)	.324812	.082529		835896	.060646		
11(π/24)	.422136	.097324	.209951	755399	.080497	.180181	
12(π/24)	.534763	.112627		655715	.099684		
13(π/24)	.663098	.128335	.272697	537338	.118377	.255075	
$14(\pi/24)$.807460	.144362		400640	.136698		
$15(\pi/24)$.968096	.160636	.337741	245901	.154739	.327305	
16(π/24)	1.145201	.177105	<u> </u>	073335	.172566		
17(π/24)	1.338926	.193725	.404188	.116897	.190232	.398006	
18(π/24)	1.549389	.210463		.324671	.207774		
$19(\pi/24)$	1.776679	.227290	.471478	.549892	.225221	.467817	
$20(\pi/24)$	2.020867	.244188	· · · · · · ·	.792488	.242596 .259915		
$21(\pi/24)$	2.282008	.261141	.539276	1.052403	.277191	.537106	
$22(\pi/24)$	2.560143	.295161		1.329594	.294435		
23(π/24)	2.855304		.607373	1.624029	.311654	.606089	
24(π/24)	3.167516	.312212		1.935683	.328853		
25(π/24)	3.496799	.329283	.675652	2.264536	.346037	.674890	
$26(\pi/24)$	3.843168	346369		2.610573	363210		
27(π/24)	4.206633	.363465	.744035	2.973783		.743585	
28(π/24)	4.587203	.380570		3.354158	.380375		
29(π/24)	4.984885	.397682	.812482	3,751689	.397531	.812215	
30(π/24)	5.399685	.414800		4.166373	.414684		
31(π/24)	5.831607	.431922	.880967	4.598205	.431832	.880808	
32(π/24)	6.280652	.449045		5.047181	.448976		
33(π/24)	6.746825	.466173	.949474	5.513301	466120	.949380	
34(π/24)	7.230126	.483301	<u> </u>	5.996561	483260		
35(π/2 4)	7.730557	.500431	1.017993	6.496961	.500400	1.017938	
36(π/24)	8.248119	.517562	<u> </u>	7.014499	.517538		
37(π/24)	8.782813	.534694	1.086521	7.549175	.534676	1.086488	
38(π/24)	9.334640	.551827		8.100987	.551812		
39(π/24)	9.903600	.568960	1.155053	8.669936	.568949	1.155034	
40(π/24)	10.489693	.586093	-	9.256021	.586085	,	
$41(\pi/24)$	11.092920	.603227	1.223588	9.859241	620356	1.223576	
4 2(π/24)	11.713281	.620361		10.479597	.620356		
43(π/24)	12.350777	.637496	1.292125	11.117089	.637492	1.292118	
44 (π/24)	13.005406	.654629		11.771715	.654626	7 .	
4 5(π/2 4)	13.677170	.671764	1.360662	12.443477	.671762	1.360658	
$46(\pi/24)$	14.366068	.688898		13.132373	.688896		
47(π/24)	15.072101	.706033	1.429200	13.838405	706032	1.429198	
48(π/24)	15.795268	.723167		14.561571	.723166	-	
49(π/24)	16.535571	.740303	1.497739	15.301873	.740302	1.497738	
		.757436	ļ		.757436		

⁽¹⁾ For negative values of $(\varphi - \Phi_O)$ the signs of α and β change, but the signs of $\Delta\alpha$ and $\Delta\beta$ remain unchanged.

Table I – Tabulated values of the integrals $~\alpha~$ and $~\beta~$ for a range of $~|(\Phi - \Phi_o)|~$ – Concluded [Computational methods given in appendix \mathbb{D} .] NACA

						a divine	
(φ -φ _o)	_α (1)	ΔΙ	= <u>∆</u> a	₆ (1)	ΔΙ =	- Δβ	
14 401	_	Δ ₁ α. (ΔΦ=π/24)	Δ ₂ α (ΔΦ=2π/24)	ŗ	Δ ₁ β (Δ Φ =π/24)	. Δ ₂ β (Δ Φ =2π/24	
(04)	15 005005	(21-1,751)	(11-11/11)	20.050700	(22-27)		
50(π/24)	17.293007	.774572		16.059309	.774571	1 = 0.00=0	
51(π/24)	18.067579	.791706	1.566278	16.833880	.791705	1.566276	
52(π/24)	18.859285	.808840		17.625585	.808841		
53(π/24)	19.668125	.825976	1.634816	18.434426	.825975	1.634816	
54(π/24)	20.494101	.843110		19.260401	.843110		
55(π/24)	21.337211	.860245	1.703355	20.103511	.860244	1.703354	
56(π/24)	22.197456	.877379		20.963755	.877380		
57(π/24)	23.074835	.894514	1.771893	21.841135	.894514	1.771894	
58(π/24)	23.969349			22,735649			
59(π/24)	24.880998	.911649	1.840433	23.647298	.911649	1.840433	
$60(\pi/24)$	25.809782	.928784		24.576082	.928784		
$61(\pi/24)$	26:755694	.945912	1,908968	25.521994	.945912	1.908968	
$62(\pi/24)$	27.718750	963056		26.485050	.963056		
63(π/24)	28.698940	.980190	.1.977507	27.465240	.980190	1.977507	
64(π/24)	29.696257	.997317		28.462557	.997317		
65(π/24)	30.710717	1.014460	2.046044	29.477017	1.014460	2.046044	
66(π/2 4)	31.742311	1.031594		30.508611	1.031594		
67(π/24)	32.791033	1.048722	2.114585	31.557333	1.048722	2.114585	
68(π/24)	33.856896	1.065863	2.111000	32.623196	1.065863	21111000	
	34.939895	1.082999	2.183123		1.082999	2.183123	
69(π/24)		1.100134	2.163123	33.706195	1.100134	2.103123	
70(π/24)	36.040029	1.117260	0.053.005	34.806329	1.117260	2.25166	
71(π/24)	37.157289	1.134403	2.251663	35.923589	1.134403		
72(π/24)	38.291692	1.151538		37.057992	1.151538		
73(π/24)	39.443230	1.168663	2.320201	38.209530	1.168663	2.320201	
74(π/24)	40.611893	1		39.378193	1.185808		
75(π/24)	41.797701	1.185808	2.388740	40.564001		2.388740	
76(π/2 4)	43.000643	1.202942		41.766943	1.202942		
77(π/24)	44.220710	1.220067	2.457279	42.987010	1.220067	2.457279	
$78(\pi/24)$	45.457922	1.237212		44.224222	1.237212		
79(π/24)	46.712269	1.254347	2.525818	45.478569	1.254347	2.525818	
80(π/24)	47.983750	1.271481		46.750050	1.271481		
81(π/24)	49.272356	1.288606	2.594357	48.038656	1.288606	2.594357	
82(π/24)	50.578107	1.305751		49.344407	1.305751		
83(π/24)	51.900992	1.322885	2.662895	50.667292	1.322885	2.662895	
84(n/24)	53.241002	1.340010		52.007302	1.340010		
85(π/24)	54.598158	1.357156	2.731435	53.364458	1.357156	2.731435	
86(π/24)	55.972447	1.374289		54.738747	1.374289		
87(π/24)	57.363861	1.391414	2.799974	56.130161	1.391414	2.799973	
88(π/24)	58.772421	1.408560	,	57.538720	1.408560	···	
89(π/24)	60.198115	1.425694	2.868512	58.964415	1.425694	2.868513	
90(π/24)	61.640933	1.442818		60.407233	1.442818	· · · · · · · · · · · · · · · · · ·	
91(π/24)	63.100897	1.459964	2.937052	61.867197	1.477098	2.937052	
92(π/24)	64.577995	1.494233		63.344295	1.494233		
93(π/24)	66.072228	1.511357	3.005590	64.838528	1.511357	3.005590	
94(π/24)	67.583585	1.528503		66.349885	1.528503		
95(π/24)	69.112088	1.545637	3.074130	67.878388	1.545637	3.074130	
96(π/24)	70.657725			69.424025	1.562761		
97(π/24)	72.220486	1.562761	3.142668	70.986786	1.579907	3.142668	
98(π/24)	73.800393	1.597042		72.566693	1.597042	· · · · · · · · · · · · · · · · · · ·	
99(π/24)	75.397435	1.614164	3.211206	74.163735	1.614164	3.211206	
100(π/24) ⁽²⁾	77.011599	T.014104		75.777899	T.014104		

⁽¹⁾ For negative values of $(\Phi - \Phi_0)$ the signs of α and β change, but the signs of $\Delta \alpha$ and $\Delta \beta$ remain unchanged. (2) For values of $|(\Phi - \Phi_0)| > 100(\pi/24)$ use equation (D2) for α and equation (D7) for β .

TABLE II - COMPARISON OF ELBOW DESIGNS OBTAINED FROM SOLUTIONS BY RELAXATION METHODS AND BY GREEN'S FUNCTION

[Linearized compressible flow; prescribed velocity distribution given in figs. 2 and $3\cdot$]

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,	l		,	-			1					$\Psi = \pi/2$				
			(Inner wa	11)				1	,	(Outer wa	11) .			
			Solution by			Solution by			Solution by					Solution by		
٠.	l		rela	axation r		Gre	een's function		rela	relaxation methods			een's fu			
•	ا و	q		(Part]			(Part I)	`	- Q	q	<u> </u>	(Part)			(Part I	`
	•	-	x	У	θ (deg)	x	У	(deg)	~	"	x	У	θ (deg)	x	. У	(deg)
00/ /04					10			+	 		 		<u> </u>			-
-22(π/24) -20(π/24)	0.5000 .5000	0.4009	-2.466 -2.241	-0.769 769	.01	-2.466 -2.241	-0.769 769	0.01	.5000	0.4009	-2.466 -2.241	.770 .770	001	-2.466 -2.241	.770	001
$-18(\pi/24)$.5000	.4009	-2.016	769	.01	-2.016	769	.01	.5000	4009	-2.016	770	01	-2.016		01
$-16(\pi/24)$.5000	.4009	-1.791	769	.02	-1.791	769	.02	.5000	.4009	-1.791	.770	02	-1.791	.770	02`
$-14(\pi/24)$.5000	.4009	-1.566	768	.03	-1.566	768	.03	•5000	.4009	-1.566	.770	03	-1.566		03
$-12(\pi/24)$ $-10(\pi/24)$.5000	.4009	-1.341 -1.116	768 768	.05	-1.341 -1.116	768 768	.05	.5000	.4009 .4009	-1.341	.769 .769	05	-1.341 -1.116	.770 .769	05
$-8(\pi/24)$.5000	.4009	891	768	.14	891	768	.14	.5000	.4009	891	.769	13	891	.769	13
$-6(\pi/24)$.5000	.4009	666	767	.24	666	767	.24	.5000	.4009	666	.768	21	666	.768	22
$-4(\pi/24)$.5000	.4009	441	766	-40	441	766	.41	-5000	.4009	441	.767	35	441	.767	36
-2(π/24)	.5000	.4009	216 .008	763 760	.70 1.31	216 .010	763	1.52	.5000	.4009 .4009	216	.765 .763	56 92	216 .009	.765 .762	59 94
2(π/24)	.5079	.4072	.233	752	2.82	.233	751	2.76	.5000	.4009	.234	.758	-1.45	.234	.758	-1.48
$4(\pi/24)$.5293	.4243	.450	739	3.88	.449	738	3.76	.5000	.4009	.459	.751	-2.22	.459	.750	-2.24
6(x/24)	.5599	.4489	.656	724	4.00	.656	724	3.89	.5000	4009	.684	.740	-3.28	.684	.740	-3.29
8(π/24) 10(π/24)	.5962 .6354	.4780 .5094	.851 1.033	712 704	3.17	.850	713	13.07	.5000	.4009	908	.725	-4.65	.908	.724	-4.68
12(π/24)	.6754	.5415	1.205	704	1.44	1.033	705 704	1.38	.5000	.4009 .4009	1.132	.703 .674	-6.41 -8.52	1.132	.702	-6.43 -8.56
$14(\pi/24)$.7149	.5732	1.366	710	-4.00	1.367	711	-4.04	.5000	.4009	1.577	.636	-11.04	1.577	.635	-11.06
$16(\pi/24)$.7531	.6038	1.519	725	-7.48	1.519	727	-7.51	.5000	.4009	1.797	.587	-13.91	1.797	.586	-13.94
$18(\pi/24)$.7894	.6329	1.663	749	-11.35	1.663	- 750	-11.37	5000	-4009	2.013	.527	-17.13	2.013	.526	-17.16
20(π/24) 22(π/24)	.8235 .8550	.6602 .6855	1.798	781	-15.52 -19.96	1.799 1.926	783 823	-15.54 -19.98	.5000	.4009 .4009	2.226	.455 .368	-20.67	2.226	.453	-20.69
24(n/24)	.8838	.7086	2.046	871	-24.62	2.046	871	-24.63	.5000	.4009	2.434	.268	-24.49 -28.58	2.434	.367 .266	-24.52 -28.59
26(x/24)	.9097	.7293	2.157	928	-29.46	2.158	929	-29.47	.5000	.4009	2.829	.153	-32.89	2.828	.152	-32.90
28(π/24)	.9326	.7477	2.261	993	-34.45	2.261	994	-34.46	.5000	.4009	3.013	.024	-37.40	3.012	.022	-37.41
30(π/24)	.9524	.7636	2.356	-1.064	-39.56	2.356	-1.066	-39.57	.5000	.4009	3.186	120	-42.08	3.185	122	-42.09
32(π/24) 34(π/24)	.9690 .9822	.7769 .7875	2.443	-1.143 -1.228	-44.76 -50.02	2.443	-1.144 -1.229	-44.77 -50.01	.5000 .5000	.4009 .4009	3.346	278 449	-46.93 -51.93	3.346 3.492	279 450	-46.94 -51.94
36(x/24)	.9919	.7953	2.590	-1.318	-55.27	2.590	-1.320	-55.27	.5000	.4009	3.623	632	-57.08	3.623	633	-57.10
38(π/24)	.9979	.8001	2.650	-1.414	-60.49	2.650	-1.415	-60.47	.5000	.4009	3.736	826	-62.44	3.736	828	-62,49
40(π/24)	1.0000	-8018	2.701	-1.514	-65.55	2.701	-1.516	-65.51	.5000	.4009	3.830	-1.030	-68.16	3.829	-1.034	-68.37
42(π/24) 44(π/24)	1.0000	.8018	2.743	-1.619 -1.726	-70.32 -74.81	2.744	-1.620 -1.727	-70.29 -74.78	.5079 .5293	.4072 .4243	3.901 3.947	-1.242 -1.455	-74.80 -81.03	3.901 3.945	-1.244 -1.456	-74.75 -80.91
46(π/24)	1.0000	.8018	2.803	-1.836	-78.97	2.803	-1.837	-78.96	.5599	.4489	3.969	-1.455	-86.33	3.969	-1.662	-86.23
$48(\pi/24)$	1.0000	.8018	2.820	-1.947	-82.82	2.821	-1.948	-82.80	.5962	.4780	3.974	-1.856	-90.69	3.975	-1.856	-90.59
50(x/24)	1.0000	.8018	2.831	-2.059	-86.28	2.831	-2.059	-86.26	.6354	.5094	3.966	-2.038	-94.16	3.967	-2.040	-94.09
52(π/24) 54(π/24)	1.0000	.8018	2.835 2.834	-2.171 -2.283	-89.37 -92.07	2.836 2.834	-2.172 -2.285	-89.34 -92.04	.6754 .7149	.5415 .5732	3.950 3.927	-2.209 -2.369	-96.93 -99.11	3.951 3.928	-2.210 -2.371	-9.6.88 -99.07
56(n/24)	1.0000	.8018	2.827	-2.396	-94.40	2.828	-2.397	-94.37	.7531	-6038	3.900	-2.520	-100.83	3.920	-2.522	-100.80
58(x/24)	1.0000	.8018	2.817	-2.508	-96.39	2.817	-2.509	-96.36	.7894	.6329	3.871	-2.663	-102.17	3.872	-2.665	-102.14
$60(\pi/24)$	1.0000	.8018	2.803	-2.619	-98.05	2.803	-2.620	-98.03	.8235	.6602	3.841	-2.799	-103.19	3.842	-2.801	-103.17
62(π/24) 64(π/24)	1.0000	.8018 .8018	2.786	-2.731 -2.841	-99.44 -100.56	2.786 2.767	-2.732 -2.842	-99.42 -100.55	.8550 .8838	.6855 .7086	3.809	-2.929 -3.055	-103.95 -104.49	3.810	-2.931 -3.056	-103.94 -104.48
66(\(\pi/24\)	1.0000	.8018	2.744	-2.952	-101.47	2.745	-2.953	-101.45	.9097	.7293	3.746	-3.176	-104.49	3.746	-3.178	-104.48
68(x/24)	1.0000	.8018	2.721	-3.062	-102.18	2.722	-3.063	-102.17	.9326	.7477	3.714	-3.294	-105.03	3.715	-3.296	-105.03
70(π/24)	1.0000	.8018	2.697	-3.172	-102.73	2.698	-3.173	-102.72	.9524	.7636	3.683	-3.409	-105.10	3.684	-3.411	-105.09
72(π/24) 74(π/24)	1.0000	.8018	2.672	-3.281 -3.391	-103.14 -103.45	2.672	-3.283 -3.392	-103.14 -103.44	.9690 .9822	.7769 .7875	3.653 3.623	-3.522 -3.633	-105.05 -104.91	3.654	-3.524	-105.04 -104.91
76(π/24)	1.0000	.8018	2.620	-3.500	-103.45	2.620	-3.501	-103.44	.9822	.7953	3.594	-3.744	-104.91	3.624 3.595	-3.635 -3.746	-104.91
78(x/24)	1.0000	8018	2.593	-3.609	-103.81	2.594	-3.611	-103.79	.9979	.8001	3.565	-3.853	-104.49	3.566	-3.855	-104.49
80(π/24)	1.0000	.8018	2.566	-3.719	-103.90	2.567	-3.720	-103.90	1.0000	.8018	3.537	-3.962	-104.28	3.538	-3.964	-104.30
82(x/24)	1.0000	.8018	2.539	-3.828	-103.97	2.540	-3.829	-103.97	1.0000	.8018	3.510	-4.071	-104.18	3.510	-4.073	-104.19

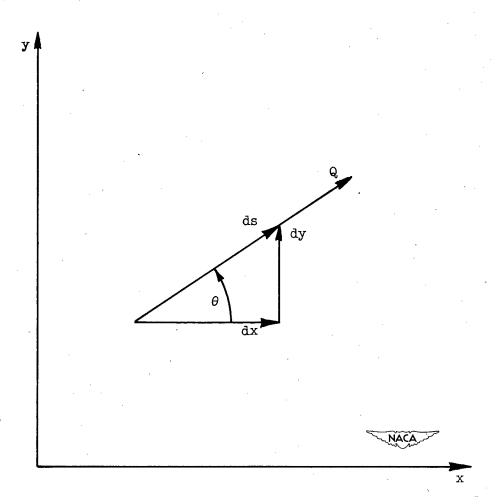


Figure 1. - Magnitude and direction of velocity at point in xy-plane.

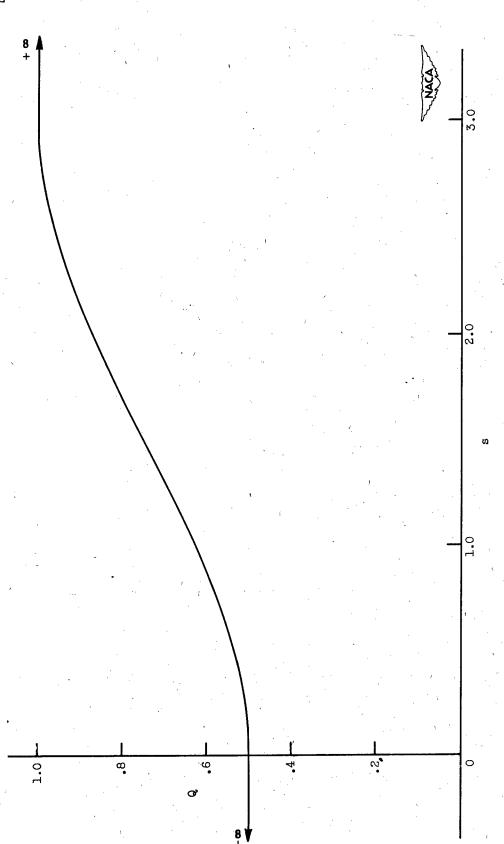
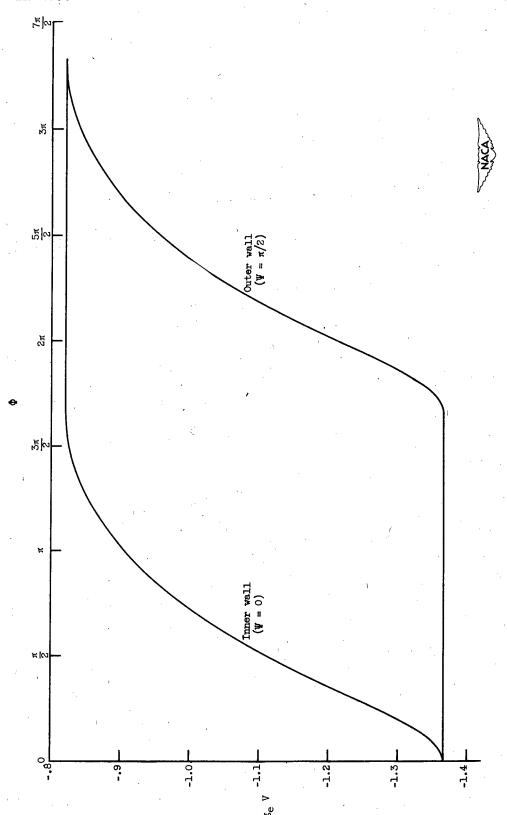


Figure 2. - Prescribed velocity distribution as function of arc length along channel wall (equation (19)).

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along channel walls of numerical example.

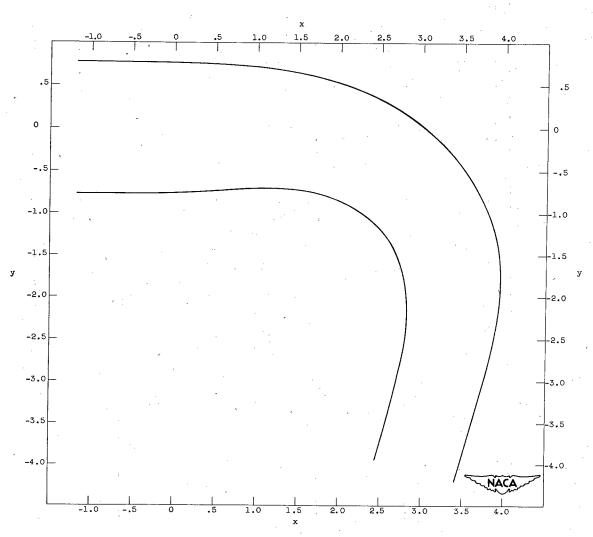


Figure 4. - Elbow design for prescribed velocity along channel walls given in figures 2 and 3.

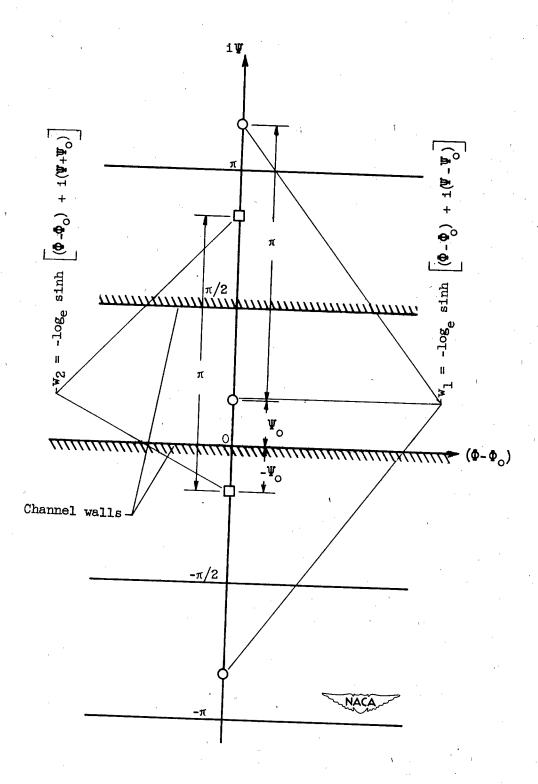


Figure 5. - Two infinite series of point sinks required in the development of Green's function of the second kind G.

1. Flow, incompressible (1.1.1) 2. Flow, Subsonic (1.1.2.1) 3. Ducts (1.4.2) I. Stanitz, John D. II. NACA TN 2595	1. Flow, Incompressible (1.1.1) 2. Flow, Subsonic (1.1.2.1) 3. Ducts (1.4.2) I. Stanitz, John D. II. NACA TN 2595
NACA TN 2595 National Advisory Committee for Aeronautics. DESIGN OF TWO-DIMENSIONAL CHANNELS WITH PRESCRIBED VELOCITY DISTRIBUTIONS ALONG THE CHANNEL WALLS. II - SOLUTION BY GREEN'S FUNCTION. John D. Stanitz. January 1952. 35p. diagrs., 2 tabs. (NACA TN 2595) Methods of solution by Green's function are developed for the design of two-dimensional unbranched channels with prescribed velocities as a function of arc length along the channel walls. The methods apply to incompressible and linearized compressible, nonviscous irrotational flow. One numerical example is presented for an accelerating elbow with linearized compressible flow. One numerical example is presented for an accelerating elbow with linearized compressible flow. The elbow shape obtained from the solution by Green's function is the same as that obtained from a solution by relaxation methods for the same prescribed conditions. The time required for the calculations is considerably copies obtainable from NACA, Washington (over)	NACA TN 2595 National Advisory Committee for Aeronautics. DESIGN OF TWO-DIMENSIONAL CHANNELS WITH PRESCRIBED VELOCITY DISTRIBUTIONS ALONG THE CHANNEL WALLS. II - SOLUTION BY GREEN'S FUNCTION. John D. Stantz. January 1952. 35p. diagrs., 2 tabs. (NACA TN 2595) Methods of solution by Green's function are developed for the design of two-dimensional unbranched channels with prescribed velocities as a function of arc length along the channel walls. The methods apply to incompressible and linearized compressible, nonviscous irrotational flow. One numerical example is presented for an accelerating elbow with linearized compressible flow. The elbow shape obtained from the solution by Green's function is the same as that obtained from a solution by relaxation methods for the same prescribed conditions. The time required for the calculations is considerably copies obtainable from NACA, Washington (over)
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